Cable Resistance in Spacecraft Deployable Mechanisms

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Introduction

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- Objective: Provide the space engineering community with a reliable and methodical way to predict these forces early in the mechanism design process
- Approach: Layout a methodology to characterize cables and record their torque response in deployment systems



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Why Does This Matter?

- Resistive torque defined late in the design cycle
- High margins are applied to account for uncertainty
- Cables experience creep effects from long term storage



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Methodology

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Cantilever Bending Experiment



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Cantilever Bending Experimental Results

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20 AWG PVC Wire (50 mm length)

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Analytical Cable Bending Model

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Analytical model alongside an optimization algorithm determined best-fit effective material properties



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Analytical Cable Model Assumptions

- Cantilevered cable is discretized into a set of elements
- Shear and axial deformations neglected



- Each element assumed to be in pure bending
- Analytical model produces results within 1% of FEA simulation



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Cantilever Results



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Constitutive Material Model

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Material model approximates how the cable deforms



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Viscoelastic Cantilever Results



Deflection held for 40 hours

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Two-Panel Representative Deployment Test

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Two-Panel FEA Model





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Two-Panel Results



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Comparison of Hybrid and Elastoplastic Model

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Conclusion

- Methodology for predicting the force response of cables in a representative deployment system
- Captured a cables bending response
- > Developed an analytical best-fit beam bending model
- Used the calibrated material properties with an FEA model to predict the response of a representative two-panel deployment system

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Future Work

- Test complex and unique cables
- Expand FEA model to other deployment setups and more complex systems
- Improve accuracy with a more robust algorithm
- Account for the thermal environment of space and longer periods of creep



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